

GPS Geodesy - LAB 4

GPS orbits

The purpose of this lab is to start familiarize yourself with GPS satellite orbits. GPS orbits are distributed in various formats, this will be covered during lectures on “Satellite orbits”. The simplest format, called “sp3”, provides the (X,Y,Z) position of the satellite center of mass in an ECEF frame every 15 minutes. This is the format used by the International GNSS Service.

(SP3 format: <ftp://igsceb.jpl.nasa.gov/igsceb/data/format/sp3.txt>), see also

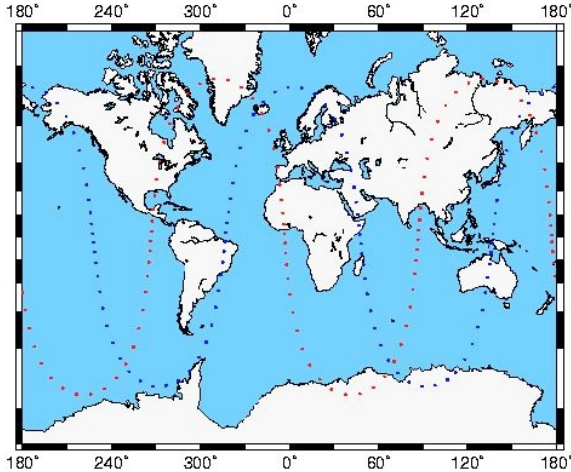
<http://www.nima.mil/GandG/sathtml/sp3format.html> and

http://www.ngs.noaa.gov/GPS/SP3_format.html)

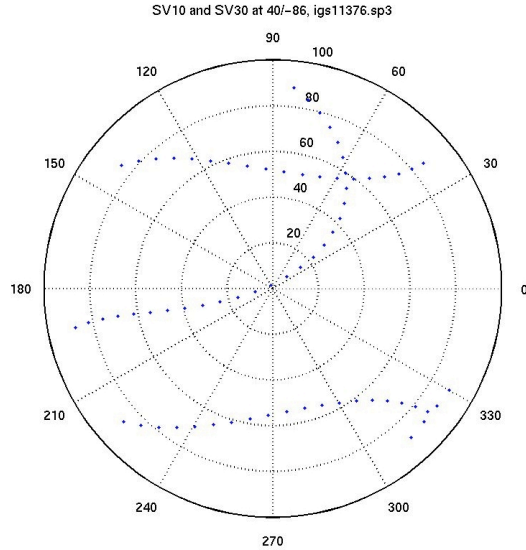
<pre>#aP1999 1 6 17 15 .00000000 151 D ITR96 FIT NOA ## 0991 321300.00000000 900.00000000 51184 .7187500000000 + 27 1 2 3 4 5 6 7 8 9 10 13 14 15 16 17 18 19 + 21 22 23 24 25 26 27 29 30 31 0 0 0 0 0 0 0 + 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 + 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 ++ 4 5 5 5 5 5 5 5 4 4 5 5 4 5 5 4 5 ++ 4 5 5 5 5 5 5 5 5 5 0 0 0 0 0 0 0 ++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 ++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 ++ 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 %c CC CC CCC CCC CCCC CCCC CCCC CCCC CCCCC CCCCC CCCCC CCCCC %c CC CC CCC CCC CCCC CCCC CCCC CCCC CCCCC CCCCC CCCCC CCCCC %f .00000000 .000000000 .00000000000 .000000000000000 %f .00000000 .000000000 .00000000000 .000000000000000 %i 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 %i 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 /* CCC /* CCC /* CCC /* CCC * 1999 1 6 17 15 .00000000 P 1 20104.806030 -13217.390413 -11082.798291 70.501167 P 2 -16426.085225 846.801170 -20466.982765 -9.683821 P 3 12408.902332 -23096.398021 -4392.698280 82.115662 P 4 -21814.685761 6583.286004 13903.675494 397.078225 P 5 11483.578489 23992.934989 -103.222680 179.645035 P 6 13914.603552 7028.569760 21655.583624 -.379980 P 7 -19497.644080 13513.068456 -11379.499312 804.727473 P 8 21255.177904 11180.772732 -11537.683401 652.383494 P 9 10922.667891 15086.394860 -19182.925877 -25.699309 P 10 -2909.051798 20868.604858 16098.215646 30.825520 P 13 -19644.179444 14713.190738 10164.040025 -431.626805 P 14 -12173.295431 -23145.973755 4512.270804 11.343876 P 15 -11729.204609 -14814.730123 -18890.726846 638.694491 P 16 -16504.576939 -14398.826376 15118.107482 96.021738 P 17 25744.646319 7081.585610 2368.464782 -199.669427 P 18 -14967.040976 -5045.055314 21129.413198 20.485541 P 19 -24482.823759 -7103.441318 7864.777634 -93.916348 P 21 14574.553758 -3853.965998 -21700.618494 2.745205 P 22 6282.098696 -17831.093585 18292.188311 607.655490 P 23 16804.363450 9819.108835 -17799.413543 8.263823 P 24 -8324.122868 13303.010877 21686.577866 465.370012 P 25 21041.028738 -8698.502930 13618.533056 .176293 P 26 -2867.387168 21433.101570 -15024.810125 746.988810 P 27 -26293.544224 1704.131282 -5324.491666 7.363154 P 29 19065.846736 -15843.465371 -9156.826537 84.477396 P 30 17020.411582 15981.210745 12498.451275 -3.471081 P 31 2229.792497 -21612.934676 -15257.431585 11.933700</pre>	<p>Header</p> <p>Time, sampling interval, reference frame, list of satellites, etc... (+ several unused fields)</p>
<pre>P 1 20104.806030 -13217.390413 -11082.798291 70.501167 P 2 -16426.085225 846.801170 -20466.982765 -9.683821 P 3 12408.902332 -23096.398021 -4392.698280 82.115662 P 4 -21814.685761 6583.286004 13903.675494 397.078225 P 5 11483.578489 23992.934989 -103.222680 179.645035 P 6 13914.603552 7028.569760 21655.583624 -.379980 P 7 -19497.644080 13513.068456 -11379.499312 804.727473 P 8 21255.177904 11180.772732 -11537.683401 652.383494 P 9 10922.667891 15086.394860 -19182.925877 -25.699309 P 10 -2909.051798 20868.604858 16098.215646 30.825520 P 13 -19644.179444 14713.190738 10164.040025 -431.626805 P 14 -12173.295431 -23145.973755 4512.270804 11.343876 P 15 -11729.204609 -14814.730123 -18890.726846 638.694491 P 16 -16504.576939 -14398.826376 15118.107482 96.021738 P 17 25744.646319 7081.585610 2368.464782 -199.669427 P 18 -14967.040976 -5045.055314 21129.413198 20.485541 P 19 -24482.823759 -7103.441318 7864.777634 -93.916348 P 21 14574.553758 -3853.965998 -21700.618494 2.745205 P 22 6282.098696 -17831.093585 18292.188311 607.655490 P 23 16804.363450 9819.108835 -17799.413543 8.263823 P 24 -8324.122868 13303.010877 21686.577866 465.370012 P 25 21041.028738 -8698.502930 13618.533056 .176293 P 26 -2867.387168 21433.101570 -15024.810125 746.988810 P 27 -26293.544224 1704.131282 -5324.491666 7.363154 P 29 19065.846736 -15843.465371 -9156.826537 84.477396 P 30 17020.411582 15981.210745 12498.451275 -3.471081 P 31 2229.792497 -21612.934676 -15257.431585 11.933700</pre>	<p>Date and time</p> <p>P = position</p> <p>Satellite number, X, Y, Z (in km), satellite clock correction (in microseconds)</p>

* 1999 1 6 17 30 .00000000						Same, 15 minutes later
P 1 19452.467561 -11932.168876 -13430.167048	70.502088					
P 2 -17495.696948 -1288.487879 -19607.913315	-9.687811					
...etc						

Satellite orbits are often visualized either by their “ground track” (= projection of their position on the Earth’s surface), or as a “sky plot” (= polar representation, valid at a given location).



Ground track = map of latitude, longitude of satellite.



Sky plot = polar representation of satellite elevation angle (0-90) and azimuth (0, 360) as see from a static position on Earth.

Assignment:

Produce a ground track and a sky plot, at latitude 40N, longitude –86, for satellites 10 and 30, using the sp3 orbit file provided [igs11376.sp3]:

1. Ground track:

- a. Extract satellite position information from sp3 file for SV10 and SV30.
- b. Convert XYZ ECEF satellite coordinates to ellipsoidal coordinates using the Matlab function written for a previous lab.
- c. To check your code, verify that:
 $Lat = 45.87907012$ $Lon = 4.67657211$ $Alt = 433.8770$ m
 Corresponds to:
 $X = 4433469.9438m$ $Y = 362672.7267m$ $Z = 4556211.6409$ m
- d. Plot ground tracks on a Mercator projection (-80<latitude<80). For plotting the map, you may use GMT, Matlab, or any other software you are familiar with.

2. Sky plot:

- a. Extract satellite position information from sp3 file.
- b. Compute ground station to satellite unit vector in geocentric XYZ coordinates.

- c. Convert that vector to unit vector (*i.e.* divide by range).
- d. Rotate it into a local North, East, Up frame (= local [n, e, u] topocentric frame at latitude 40N, longitude 86W) using the Matlab function written for a previous lab.
- e. Convert [n, e, u] coordinates to azimuth and elevation angle using:
Horizontal length of unit vector = $\sqrt{n^2+e^2}$
Angle from Zenith to satellite = $\text{atan2}(\text{hlen}, u)$
Azimuth from North = $\text{atan2}(e, n)$
- f. Discard data when azimuth angle is below horizon.
- g. Plot data on a polar plot.

Use any combination of csh, GMT, C, F77, or Matlab. Provide source code, executable (code only if Matlab), and resulting plots with explanation if necessary.